CH EN 230 Circle one ME EN 2300

To receive full credit on problems you must clearly indicated the equations you are using (if you use equations) and make clear your steps in simplifying the equations. Do not plug in any numbers until the last step or until necessary.

1. (15) The enthalpy of water at 100 C is 2300 kJ/kg.

a) What is the phase of the water? Explain

b) What is the phase of the water if the temperature is raised 3 C while keeping the

pressure constant. Explain

answer right 2

answer right 2

explaination cowed 6

Becomes enthalphy between the sort liquid

& scd. vapor

b) Euperheaded various 2 explanation 5

If a mixture, Temp can only rise after all liquid has been converted to vapor. So if temp rises by any amount of all,

it much be sweethersteel

2. (25) The temperature of nitric oxide gas is raised from 500 K to 1000K. Using the most accurate method at your disposal. Determine the change in internal energy, in kJ/kg of the nitric oxide for this temperature rise.

$$du = CvdT \quad most \quad accurate is polynomial expression$$

$$du = \int_{T_1}^{T_2} (Q - Ru) + bT + cT^2 + dT^3 dT \qquad R_{M} = 9.21047 \ k_{mal} k$$

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$$du = CvdT II$$

$$\Delta u = \int CvdT J Z$$

$$Cv = Cp - Ru Z$$

integral al valuate 7

Molon & mass conversion

CV = Cy 2

Mevaluation 2

Res 1

all consect 1

3. (15) Consider the mixing chamber shown below where flow enters/exits uniformly through three openings. The density is a constant. The following is given:

$$A_1 = 0.1 \text{ m}^2$$

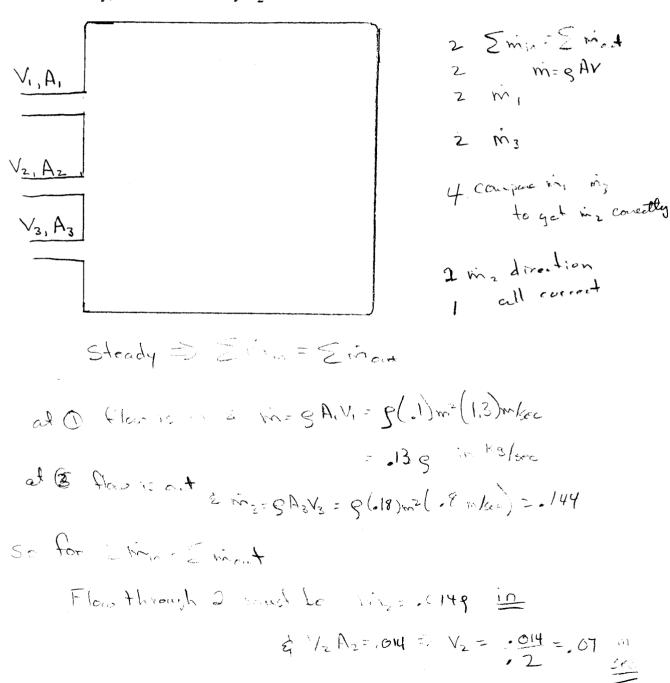
$$A_2 = 0.2 \text{ m}^2$$

$$A_3 = 0.18 \text{ m}^2$$

 $V_1 = 1.3$ m/s (into mixing chamber)

 $V_3 = -0.8 \text{ m/s}$ (out of mixing chamber)

If the flow is steady, what is the velocity V₂? Is this in or out of the control volume?

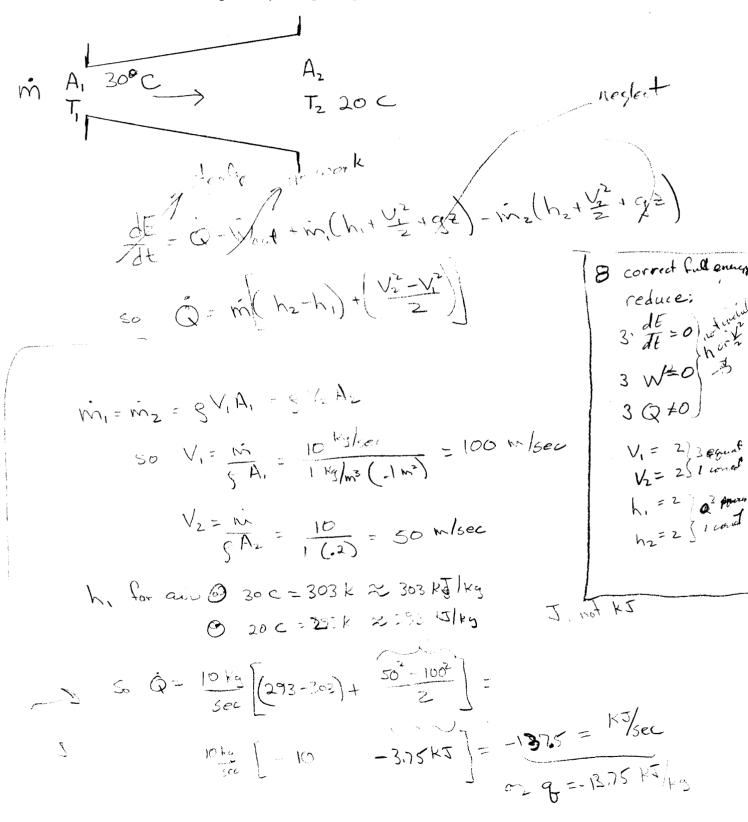


4. (20) A shock is a very thin region in a flow where velocity, temperature, pressure, density, etc. have large jumps in their values. Consider a shock formed in a pipe. There is no heat transfer, the mass flow rate is m, and the flow is steady. Draw a control volume around the shock and reduce the energy equation (1st law) to its simplest appropriate form to relate properties on either side of the shock. Since the flow is horizontal, there are no potential energy changes across the shock.

3 draw C.V. 6 correct Cull energy reduce 3 dE:0 3 Q:0 3 V:0 1st Law Cor C.V. 2 all most In Ochorale Com dE = Q-Wout + m, (h, + \frac{V_1^2}{2} + g \green_1) - m_2 (h_2 + \frac{V_2^2}{2} + g \green_2) de o because stoudy Que o because problem states no h.t. So left with m(h, + \frac{\sqrt{2}}{2}) - m_2(h_2 + \frac{\sqrt{2}}{2}) = 0 Wind = 0 no potential but in = m2 (steady) $50 \int h_1 + \frac{v_2^2}{2} = h_2 + \frac{v_2^2}{2}$ Other form O.k like (hz-hi) + (12-Vi) = 0

5) (25) Air at 30 C enters a diffuser with area $A_1 = 0.1 \text{ m}^2$. The air exits at $A_2 = 0.2 \text{ m}^2$ at a temperature of 20 C. Assume the density remains a constant value of 1.0 kg/m³ through this process. This is a steady flow problem with a mass flow rate of 10 kg/s.

A) Compute the rate of heat transfer into or out of the nearle. Indicate the direction of heat transfer. You can neglect any changes in potential energy.



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